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The Determination of Exhaust Cloud Dimensions From Films of Space Shuttle Launches

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SECTION 1 - INTRODUCTION

Photogrammetry, the technology of obtaining quantitative information about objects through the use of photographic images, has been applied to many disciplines in the past. More recent applications have evolved from remote sensing activities in space. Of particular interest is the digital processing of multispectral images from weather satellites and earth observation platforms for the purpose of stereographic measurements of three-dimensional clouds. Quantitative processing of cloud images from cameras both on the ground and aloft provide the ground truth for many meteorological and earth sciences experiments where the "laboratory" is the real atmosphere and where little, if anything, can be controlled in the usual laboratory sense. For a discussion of some of these applications see Holle, (1982).

The purpose of this report is to document cloud dimensions from 16 mm films of the Space Shuttle launches from Kennedy Space Center, Florida. The dimensions of the ground cloud formed from the rocket exhausts will be the ground truth for numerical simulations of such clouds in different atmospheric conditions.

SECTION 2 - DATA SOURCES

Motion picture films were taken from Kennedy Space Center by TGS

Technology Inc. in order to record the Shuttle exhaust cloud near the

ground. These films were taken for three Shuttle launches: Mission 41C

(April 6, 1984), Mission 41D (August 30, 1984), and Mission 51A (November 8,

1984). Kodak 7239 16 mm color film was used in Mitchell cameras with an

exposure time of 1/108 second. Different lenses were used to capture both the

full ground cloud and the horizon in each frame in order to facilitate

quantitative analysis. Copies of these films were provided for this study.

For each launch, films were taken from three different Universal Camera Sites (UCS) whose locations are described by Figure 1. The focal lengths of the camera lenses and speeds of the films are shown in Table I. Unfortunately, the film from UCS 9 for Mission 51A was overexposed and unavailable for analysis. The locations of the camera sites, launch pads, and other structures were determined from detailed maps provided by Kennedy Space Center.

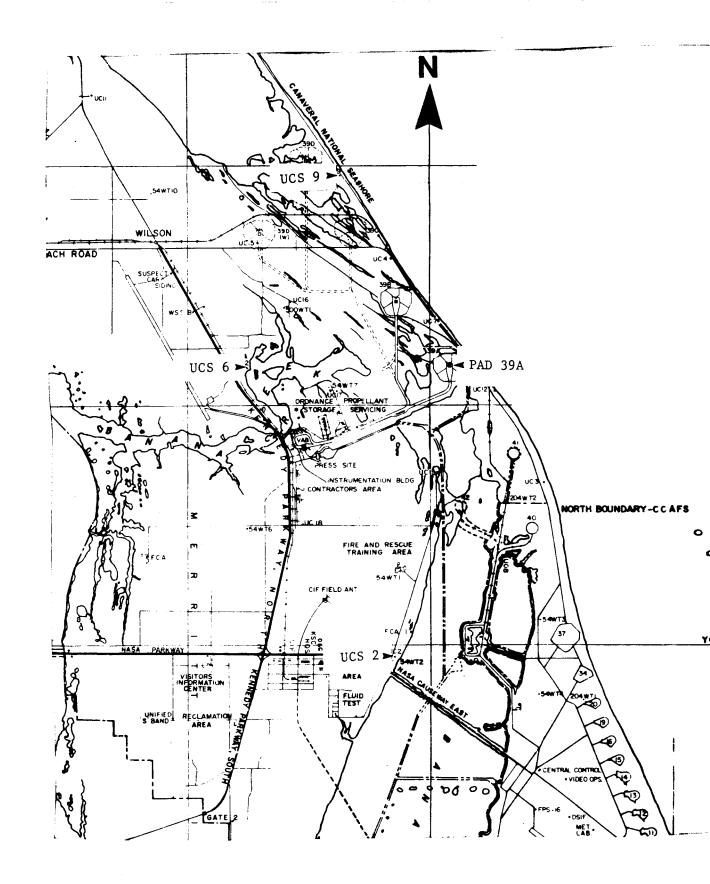


Figure 1. Locations of camera sites

TABLE 1: CAMERA LOCATIONS AND FOCAL LENGTHS

Mission	Date	Liftoff (GMT)	Camera Site	Camera Location (From Pad 39A)	Focal Length	Film Speed (Frames/Sec)
41C	4/6/84	13:58:00	UCS 2 UCS 6 UCS 9	190.6° at 9330 m 270.0 at 6320 331.2 at 6975	15 mm 15 10/15	18.6 19.5 18.7
41D	8/30/84	12:41:50	UCS 2 UCS 6 UCS 9	190.6 at 9330 270.0 at 6320 331.2 at 6975	15 10 15	3.0 3.0 3.0
51A	11/8/84	12:15:00	UCS 2 UCS 6 UCS 9	190.6 at 9330 270.0 at 6320 UNAVAILAB	15 15 LE	3.0 3.0

In order to permit reconstruction of the proper time history of the Shuttle launches and ground clouds, each film was imprinted with a special NASA-developed timing code known as IRIG Format 'B'. The films with 20 frames/second speed were coded with 100 pulse per second (pps) IRIG code, while the films at 3 frames/second speed were coded with 10 pps code. The formats of the two codes are identical except that the 10 pps code does not include the straight binary time representation at the end of each sequence and the pulse sequences repeat every 10 seconds instead of each second (see Figure 2). The time indicated by each pulse sequence corresponds to the frame nine inches (30 frames) behind the point on the film where the first pulse of that sequence is printed.

Figure 2. IRIG Code

2-4

SECTION 3 - METHODS

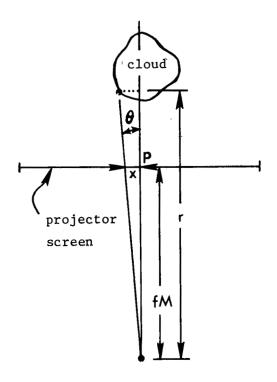
Outlines of the cloud were traced onto paper for subsequent analysis using a 16 mm stop-action projector. The projector magnification (ratio of projector's screen dimensions to those of the film) was adjusted to be 83.4 for all tracings. At one-minute intervals for each film, the cloud outline was traced along with the screen's Principal Point (the exact center of the frame), the horizon, and any ground features such as launch assemblies for Pads 39A, 39B and buildings or towers. In addition, the positions of the Shuttle itself were noted for each film at 5-second intervals while the vehicle was within the camera's field of view; these Shuttle positions were used to check equations for range and height by comparing the calculated heights to altitudes on detailed Ascent Trajectory Listings from Johnson Space Flight Center.

Ground-Based Photogrammetry

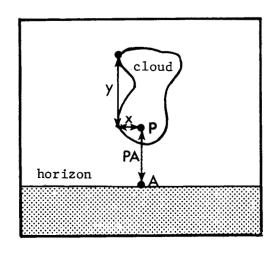
Heights and widths of objects on the projector screen can be converted to actual distances. To locate a point on the exhaust cloud, it is necessary to calculate two angles (see Figure 3): θ , the horizontal angle (projected onto the Earth's surface plane) from the Principal Point 'P' to the cloud point with the camera site as the center; and σ , the vertical angle from the ground to the cloud point.

The equations to determine these angles are (Holle, 1982):

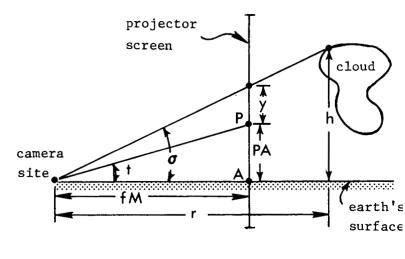
$$\tan(\theta) = \frac{x}{fM\cos(t) - y\sin(t)}$$
 (1)



AERIAL (TOP) VIEW



PROJECTOR (FRONT) VIEW



SIDE VIEW

Figure 3. Variables used in photogrammetry equations

and

$$tan(\sigma) = \frac{\cos(\theta)[y\cos(t) + fM\sin(t)]}{fM\cos(t) - y\sin(t)}$$
 (2)

where 't' is the camera tilt (with respect to horizontal) given by

$$tan(t) = -\frac{PA}{fM}$$
 (3)

and the other parameters (which must all have the same units) are indicated on Figure 3 and defined in Table 2.

TABLE 2: PARAMETERS USED IN PHOTOGRAMMETRY EQUATIONS

Parameter	Description
PA	Vertical distance from Principal Point to horizon (measured on projector screen)
f	Focal length of camera
м	Magnification of the projector
x	Horizontal distance from Principal Point to cloud point on projector screen
у	Vertical distance from Principal Point to cloud point on projector screen

Once the angles θ and σ have been calculated, the height (h) of a cloud point can be determined from

$$h = rtan(\sigma)$$
 (4)

where 'r' is the range (ground distance) to the cloud point from the camera site. In addition, the width (w) between two cloud points can be calculated from

$$w = r tan(\theta_2 - \theta_1)$$
 (5)

Cloud Dimensions

Four different quantities were calculated for the exhaust cloud in each film: the altitudes of the top and base, the maximum width (near the top of the cloud), and the average width. Figure 4 illustrates the greater width of the ground cloud near the top and the irregular shape of the cloud in general. The altitude of the top was considered to be the highest point on the cloud. The height of the base, however, was selected according to where the cloud material appeared to become more dense and better defined. The smaller "pieces" of the cloud that protruded beneath the dense areas were not considered in defining the cloud base. The maximum width was specified to be near the top of the cloud in order to exclude from consideration the inactive elements near the base which frequently extended beyond the main body of the cloud. The average width was estimated to give a general horizontal dimension to the cloud despite its non-uniform shape.

Cloud Ranges

When the range 'r' is known for one camera site, the cloud dimensions can be calculated from that film. However, for the Shuttle exhaust cloud, the range from one camera site must be determined from at least two different

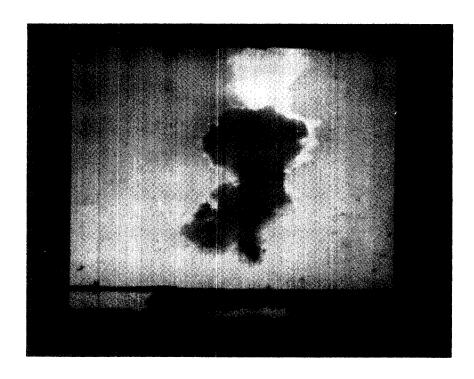


Figure 4. Photograph of a 512 x 512 digitization of a film frame for Mission 41D at nine minutes after liftoff. This picture was taken from camera site UCS-6 looking east.

films for each launch. Since the launch assembly on Pad 39A (from which the Shuttle was launched in each case) was visible in films taken from UCS 6 and UCS 9, the cloud's distance from Pad 39A in these two views was used to triangulate the cloud's position. Since the UCS 9 film was not available for Mission 51A, a straight line approximation was employed to represent the cloud's movement and to calculate its ranges from the camera sites of the other two films. On all films taken from UCS 2, the horizon, launch assemblies, and other ground objects are obscured by foliage near that camera site; these films could not be used to triangulate positions of the clouds.

Ranges for Missions 41C, 41D

The coordinates of the exhaust cloud with respect to UCS 6 $(r_6, {}^{\phi}_6)$ and UCS 9 $(r_9, {}^{\phi}_9)$ were determined in two steps: (1) the angles ${}^{\phi}_6$ and ${}^{\phi}_9$ were determined from quantities measured on the projector screen at each selected time throughout the cloud's duration; and (2) the ranges r_6 and r_9 were calculated based on these angles and the locations of the camera sites and Pad 39A.

From Figure 5, the angles ϕ_6 and ϕ_9 are determined by

$$\phi_6 = \arctan(1_6/fM_6) + \arctan(c_6/fM_6)$$
 (6)

and

$$\phi_9 = \arctan(1_9/fM_9) + \arctan(c_9/fM_9) + 28.8^{\circ}$$
 (7)

The parameters in these equations are described in Table 3.

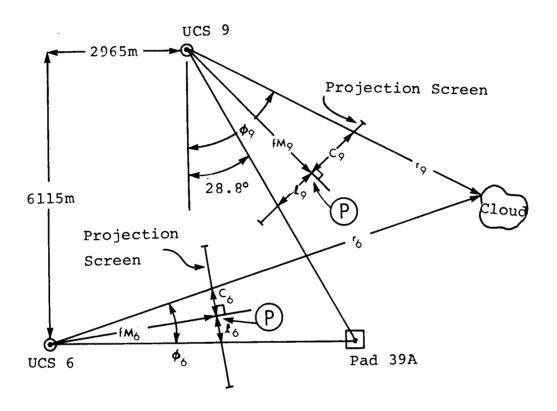


Figure 5. Ranges and azimuths of exhaust cloud from camera sites

TABLE 3: PARAMETERS USED IN RANGE EQUATIONS (6) and (7)

Parameter	Description
1 ₆ and 1 ₉	Distance on the projector screen from Principal Point 'P' to Pad 39A, defined as positive when Pad 39A is to the right of 'P'
c ₆ and c ₉	Distance on the projector screen from 'P' to the cloud point, defined as positive when cloud point is to the left of 'P'
fM ₆ and fM ₉	Products of the focal lengths and magnification factors defined previously

Once the angles ϕ_6 and ϕ_9 have been computed, the ranges can be found by equating the following distances on Figure 5:

East-West distance from UCS 6 to cloud: $r_6 \cos \phi_6 = r_9 \sin \phi_9 + 2965$

North-South distance from UCS 6 to cloud: $r_6 \sin \phi_6 = 6115 - r_9 \cos \phi_9$ (the distance unit is meters)

Solving for r_6 and r_9 yields

$$r_6 = \frac{(2965) + (6115)\tan\phi_9}{\cos\phi_6 + \sin\phi_6 \tan\phi_9}$$
 (8)

and

$$r_9 = \frac{(6115) - (2965)\tan\phi_6}{\cos\phi_9 + \sin\phi_9 \tan\phi_6}$$
 (9)

Ranges for Mission 51A

Since the UCS 9 film was unavailable for Mission 51A, a different method was used to calculate the cloud ranges. The UCS 2 and UCS 6 films showed the

cloud to be above the Vertical Assembly Building (VAB) at T + 10 minutes. The cloud's movement was approximated to be along a straight line from PAD 39A to the building. Since the range of the building is approximately 2940 m from UCS 6 and 7135 m from UCS 2, the following equations were developed to describe the cloud's range with time from each camera site:

$$r_6 = 6320 + \frac{2940 - 6320}{10 \text{ min.}} (T) = 6320 - 338(T)$$
 (10)

$$r_2 = 9330 + \frac{7135 - 9330}{10 \text{ min.}} (T) = 9330 - 219.5(T)$$
 (11)

SECTION 4 - METHOD CONFIRMATIONS

The equations used to calculate cloud ranges and heights were tested on objects visible on the films whose positions and dimensions were known from other sources. For example, the ranges to the launch assembly of Pad 39B (which is near Pad 39A) were calculated and compared to distances measured on detailed maps from Kennedy Space Center. In addition, the photogrammetry equations were used to calculate the height of the Verticle Assembly Building as well as the altitude of the Shuttle during its first 30 seconds of flight. The calculated height of the VAB was compared to that given by construction drawings (furnished by Kennedy Space Center), and the Shuttle heights were compared to Ascent Trajectory Listings from Johnson Space Flight Center.

Range Comparisons

The range equations, Eqns. (6) through (9), were used to find the distances from Pad 39B to UCS 6 and UCS 9 for Mission 41D at T + 5:00 minutes. The variables c_6 and c_9 represent the horizontal distance from 'P' to the launch assembly of Pad 39B instead of to a cloud point. The data and results of this comparison are shown in Table 4.

TABLE 4: COMPARISON OF RANGES TO PAD 39B: CALCULATED VALUES VS. MAP MEASUREMENTS

Camera Site	Cloud 1	Data c	(cm) fM	Calculated \$\phi\$ (deg.)	Quantities Range (m)	Map Ranges	Error
UCS 6	5.6	31.8	83.4	24.7	5120	5120	0
UCS 9	8.3	-21.1	125.1	23.0	4320	4390	1.6%

From both cameras, the calculated ranges are very close to those measured from maps.

Photogrammetry Comparisons

The photogrammetry equations, Eqns. (1) through (5), were used to calculate the height of the VAB (which appears at T + 8:00 minutes on UCS 6 and UCS 2 films for Mission 51A). The height calculated in each view was compared to that specified on a drawing containing the building plans. The results of this comparison are shown in Table 5.

TABLE 5: COMPARISON OF HEIGHTS OF VAB: CALCULATED vs. ACTUAL (FROM ENGINEERING DRAWING)

	С	loud Data	(cm)		Ca	alculate	d Quant:	ities		Actual	Error
Camera Site	PA	fM	x	У	t ^o	θ ο	tan	r(m)	h(m)	Height (m)	i.
UCS 6 UCS 2	26.4 27.4	125.1 125.1	0.0 40.6	-19.8 -24.4	11.9 12.4	0.0 17.7	.0511	2940 7135	150 159	160 160	6.2%

The photogrammetry equations were also used to calculate the altitude of the Space Shuttle during its first 30 seconds of flight for Mission 41C. The range equations were used to locate the Shuttle as it moves away from the camera sites. The calculated heights were compared to one-second Ascent Trajectory Listings from Johnson Space Center (JSC). The results of these comparisons are shown in Table 6.

TABLE 6: COMPARISON OF SHUTTLE ALTITUDES: CALCULATED VS. DATA FROM ASCENT TRAJECTORY LISTINGS

Time (s)	JSC Data	UCS 2	UCS 6	ucs 9	Ave. Error
T + 15 T + 20 T + 25 T + 30	670 m 1241 2000 2940	715 m 1285 2025 3005	765 m 1335 2130	750 m 1325 2120 3090	10.4 % 6.0 4.5 3.6

In all cases, the heights calculated from the equations agree well with reliable ground truth data.

SECTION 5 - RESULTS

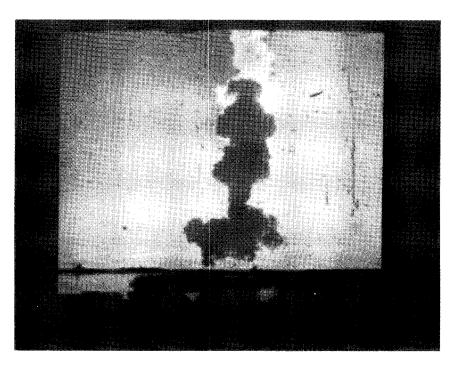
From Equations (1) through (11), the altitudes of the top and base and the maximum and average widths of the Shuttle exhaust cloud were calculated for each film tracing at one-minute intervals. Since the shapes of the sides and bases of the cloud were irregular, the altitudes of the base and average widths were estimated and indicated on each tracing. In general, the ground cloud develops from the combination of portions of the exhausts coming from the vehicle itself and the flame trenches which are separated by 180 degree and into which deluge water is sprayed for cooling. The cloud rises to its maximum altitude depending on atmospheric conditions. For these three launches the cloud reaches its maximum altitude in 3 to 5 minutes then descends slightly due to the influence of the environment. 1

The Shuttle turns East just after launch, and the ground cloud tended to follow this tilted exhaust channel as shown in Figure 6a. Notice the irregular shape. Contributing to this asymmetry were wind shear (really only for Launch 51A), the tilt of the rocket exhausts as the Shuttle turned East, and the three channels for exhaust near the ground (rocket engines after they clear the pad, and the two pieces from the flame trenches). These pieces are somewhat visible from the films in the early few minutes of cloud formation (Figure 6b). The top of the cloud in Figure 6b is tilted into the picture and corresponds to the left (Eastward) tilt in Figure 6a. The data for all cloud measurements are presented in Appendix A.

¹A temperature inversion was present just below the maximum altitude; this creates a situation whereby cloud parcels cooling as they rise, reach a temperature which is colder than the environment whereby these parcels descend to an equilibrium temperature near the level of the inversion.



(a) UCS 9 View (looking South)



(b) UCS 6 View (looking East)

Figure 6

Photographs of 512 x 512 digitizations for Mission 41D at three minutes after liftoff.

Results For Mission 41C

The calculated cloud top and base are shown for UCS 6 and UCS 9 films in Figure (7a). Measurement from these camera views showed the altitude of the cloud top to reach a peak of 2200 m at T + 4:00 minutes followed by a rapid decline to 1700 m by T + 6:00 minutes. The top remained at approximately 1700 m until the cloud began to dissipate after T + 9:00. The base of the cloud rose steadily after T + 1:00 and approached an asymptote of 1000 m at T + 10:00. The altitudes calculated from the two different camera views agreed to within 160 m for the cloud top and to within 110 m for the base.

The calculated maximum cloud widths near the top and the average widths from UCS 6, UCS 9, and UCS 2 (Fig. 7b) show the cloud to be quite asymmetrical. The cloud appeared to be much wider after T + 6:00 in the North-South direction (UCS 6) than in the East-West direction (UCS 2, 9). The maximum width near the top as measured from UCS 6 increased almost linearly with time to 2500 m at T + 10:00, while that measured from UCS 2 reached a peak of 1800 m at T + 7:00 and subsequently decreased. The maximum width from the UCS 9 view peaked at 1500 m at T + 4:00 and remained between 1300 m and 1500 m afterwards. Figure (7b) indicates that the cloud's orientation may have shifted between T + 6:00 and T + 9:00 possibly as a result of a change in wind directions. Figure (7c) shows the average width measured from UCS 6 after T + 6:00 to be about 800 m greater than the stable 1200 m width measured from both the UCS 2 and UCS 9 views.

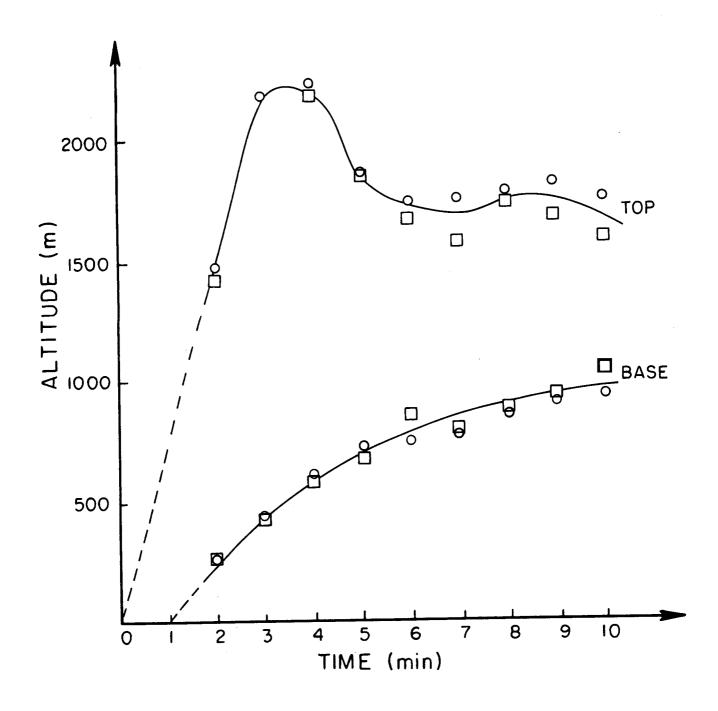


Figure 7a. Altitude of top and base of Shuttle exhaust cloud vs. time Mission 410

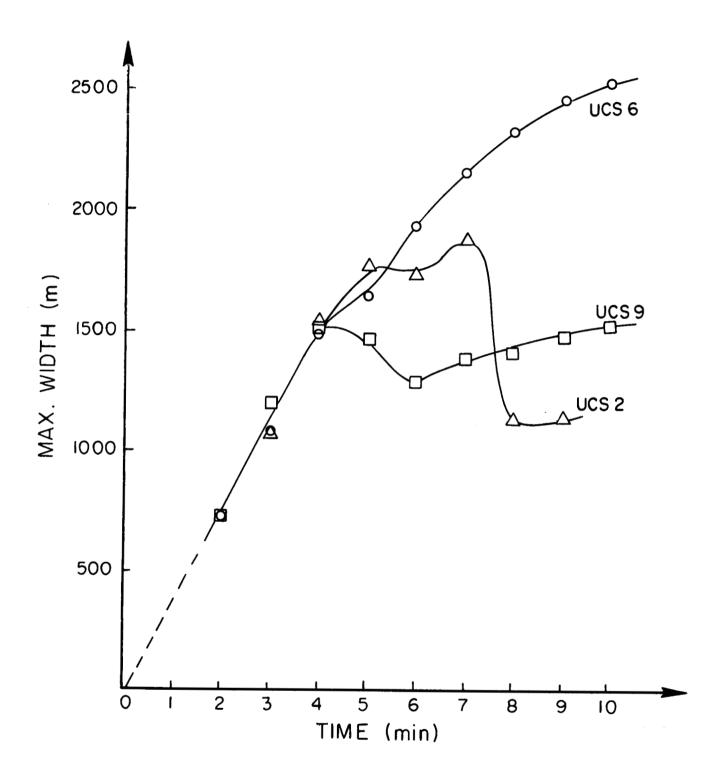


Figure 7b. Maximum widths (at top) of Shuttle exhaust cloud vs. time Mission 41C

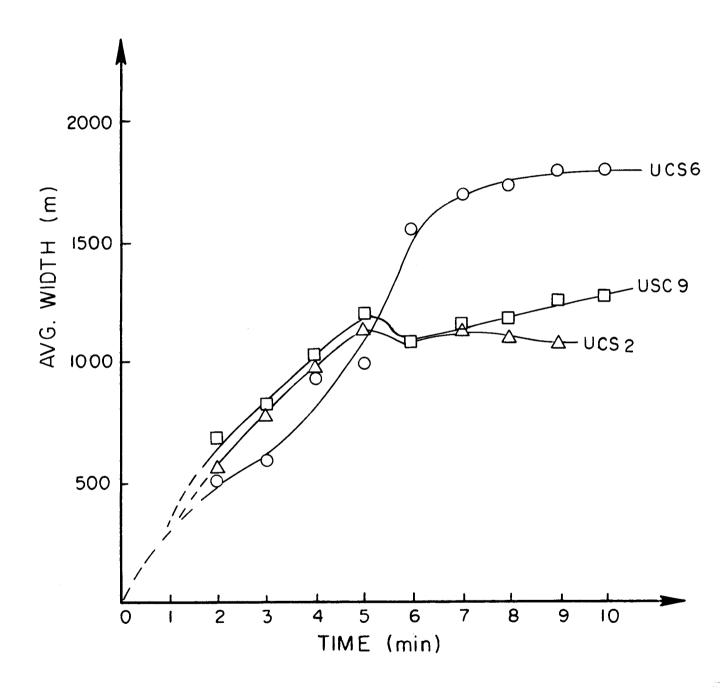


Figure 7c. Average widths of Shuttle exhaust cloud vs. time $$\operatorname{Mission}$$ 41C

Results For Mission 41D

The cloud top and base for Mission 41D were calculated from UCS 6 and UCS 9 films and are shown in Figure (8a). Measurements from these camera views showed the altitude of the cloud top to reach a peak of 3500 m at T + 5:00 followed by a gradual decline to 3000 m at T + 10:00. In general, the cloud top for Mission 41D rose to a much higher altitude and remained there for a longer time than that of Mission 41C. Like that of Mission 41C, however, the base of the cloud for 41D rose steadily after T + 1:00 and reached 1000 m at T + 10:00. The altitudes calculated from the two different camera views agreed to within 190 m for the cloud top and to within 100 m for the base, which was difficult to estimate for most of the tracings because of its non-uniformity.

The maximum widths (near the cloud top) and the average widths are shown in Figure 8b. The calculated maximum cloud widths from UCS 6 and UCS 9 show the cloud to be quite asymmetrical near the top. For this launch, the cloud was up to 700 m wider in the East-West direction (UCS 9) than in the North-South direction (UCS 6) between T + 4:00 and T + 7:00. At the other times, however, the view-to-view width difference was less than 200 m. The maximum width near the top measured from both UCS 6 and UCS 9 rose asymptotically towards 1200 m at T + 10:00.

The volume was estimated for the ground cloud at 5 minutes after launch. At this time the first convective element was at its maximum altitude (which was also the absolute maximum) but the main bubble was still rising and the cloud from the North trench was dissipating. The extemely irregular shape of the cloud, the fact that different convective elements were rising or dissipating at different rates and that these convective elements were

overlapping to varying extents makes any precise volume calculations impossible using only ground-based photographs. Nevertheless geometric approximations were made to represent the cloud at the 5 minute point. Two methods were used. In the first a rectangular solid represented the bottom part with dimensions of 1548 m x 1305 m x 840 m. A cylinder of radius 525 m and height 1627 m was the central case and a half sphere of radius 520 m was the top. The volume was calculated to be $3.37 \times 10^9 \text{ m}^3$ for this method. In method 2 an overlapping and offset to the North sphere of radius 507 m was the base. A cylinder of 525 m radius and height of 1776 m was the center with an overlapping sphere of radius 581 m to represent the main convective bubble still rising but below the top bubble by about 800 m. The top was a half sphere of radius 520 m the same as in method 1. The volume this time was $3.05 \times 10^9 \text{ m}^3$. These numbers turned out to be surprisingly close despite the difference in assumed shapes.

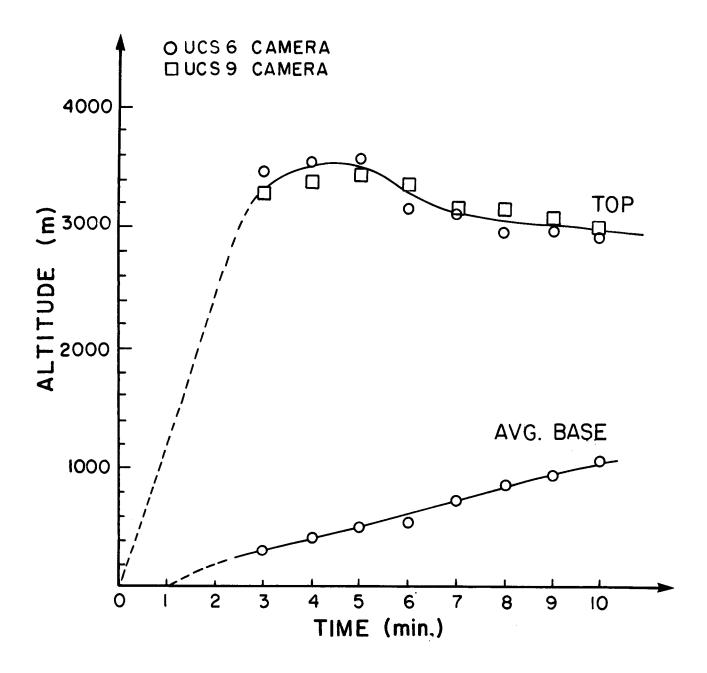


Figure 8a. Altitude of top and base of Shuttle exhaust cloud vs. time Mission 41D

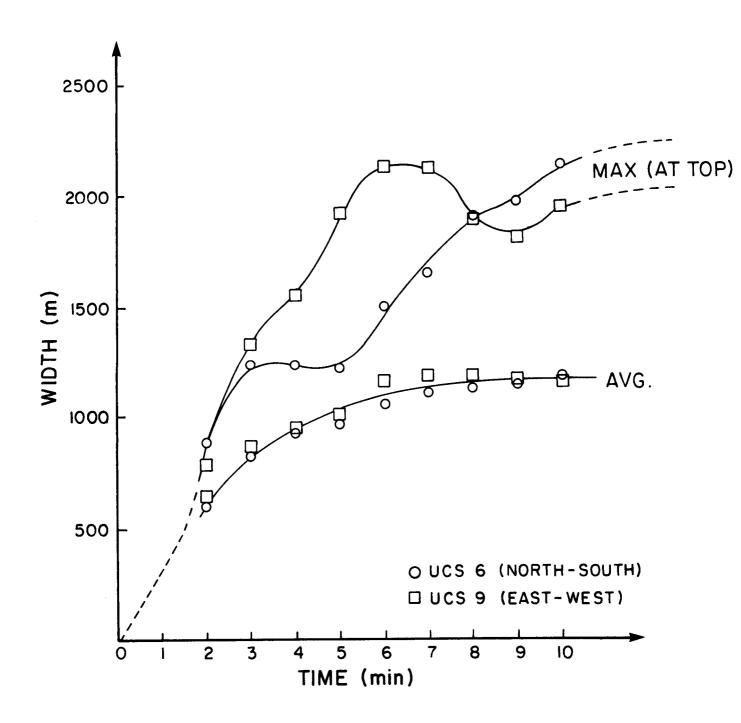


Figure 8b. Average widths and maximum widths (at top) of Shuttle exhaust cloud vs. time Mission 41D

Results For Mission 51A

For Mission 51A, the UCS 9 film was unavailable for this study. In addition, much of the cloud was obscured by environmental clouds in the UCS 2 view. Consequently, most cloud measurements were forced to be taken from the UCS 6 film alone. Only the average width could be estimated from the UCS 2 view based on the lower, visible portion of the cloud.

The calculated cloud top and base are shown for the UCS 6 films in Figure 9a. The altitude of the cloud top was estimated to reach a peak of about 2400 m at T + 4:00, since the cloud extended above the camera's view. After T + 5:00, the cloud top fell rapidly to 1800 m by T + 8:00. The cloud base, similar to those of the other launches, rose steadily after T + 1:00 and reached about 900 m at T + 8:00. By T + 8:00 after launch, the cloud had spread out considerably and moved too close to the camera site to be entirely contained within picture frames.

The average and maximum cloud widths for Mission 51A are shown in Figure 9b. The calculated maximum cloud width from UCS 6 reached a peak of 1700 m at T + 6:00, but was difficult to estimate afterwards. The average width was calculated from UCS 6 and estimated from UCS 2 based on what was visible beneath the environmental cloud cover. The UCS 2 calculations showed the average width to be up to 700 m wider than that from UCS 6. However, since the cloud shape was diagonal from top to bottom in the UCS 6 view, the average width was measured diagonally across the cloud (perpendicular to its sides) in order to provide a more accurate width that could be used in a rough volume estimate. As a result, the UCS 6 width may have been considerably less than that which was measured horizontally (because of the limited visibility of the cloud) from the UCS 2 view.

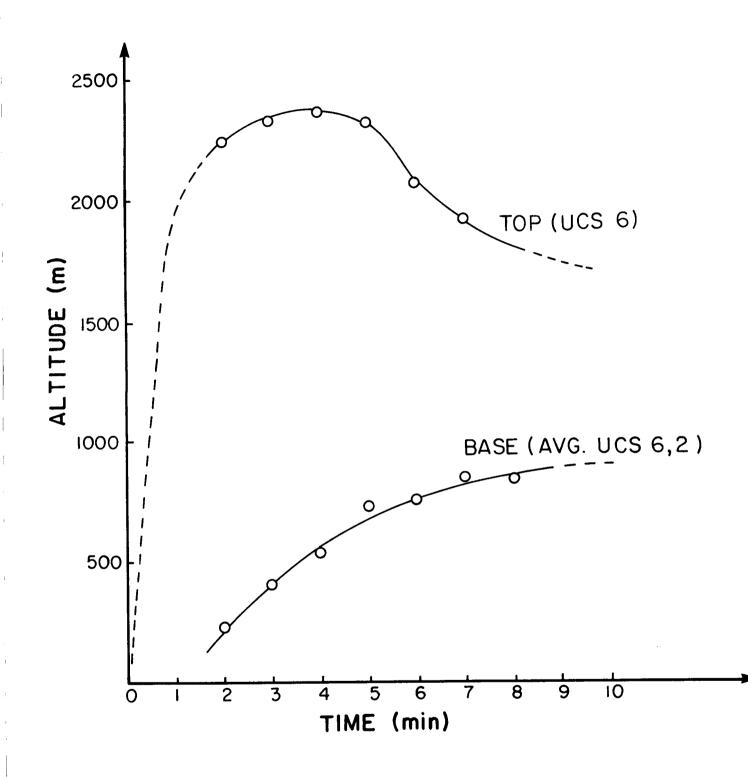


Figure 9a. Altitude of top and base of Shuttle exhaust cloud vs. time Mission 51A

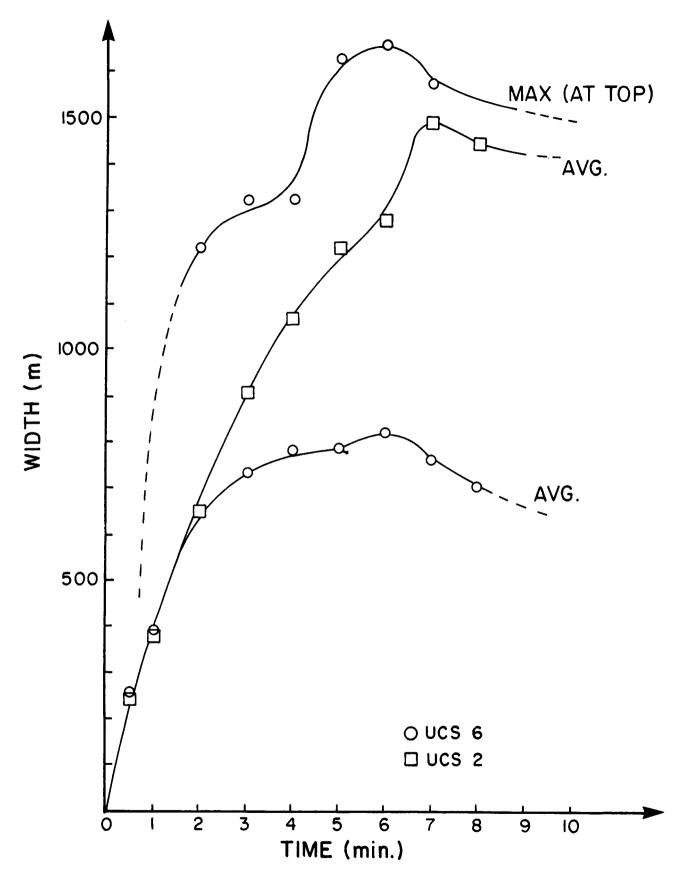


Figure 9b. Average widths and maximum widths (at top) of Shuttle exhaust cloud vs. time Mission 51A

SECTION 6 - UNCERTAINTIES AND ERROR SOURCES

The errors and uncertainties can be separated into three main categories: random errors, errors due to the limited visibility of the exhaust cloud, and errors due to the cloud's non-uniform shape. The random errors, caused by uncertainties in direct measurements, had a relatively small effect on the results of this study. The errors due to limited visibility of the cloud and its non-uniform shape were significant and caused some measurements to be estimated or abandoned. Most of the errors in heights or widths were due to these factors. Considering all sources of errors and uncertainties, the results can be considered accurate to within about 350 m.

Random Errors

The random errors in calculating the exhaust cloud dimensions were those due to uncertainties in measuring distances from maps and from the cloud tracings. Since the map distances were read to the nearest 10 m, their uncertainties had little effect on the accuracy of the cloud calculations. Likewise, the uncertainties in measuring distances from the tracings of the clouds were negligible. The effects of these uncertainties were determined by varying the measured parameters by about 3 mm (which was the approximate measuring uncertainty) in different combinations and observing the effect on ranges and cloud dimensions. The ranges and dimensions typically varied less than 100 m. The estimated positions of the horizon in the UCS 2 views can also be considered measurement uncertainties; their effects were circumvented by using the two other views for Missions 41C and 41D and were irrelevant for 51A because the cloud itself was obscured. Thus, random errors had a minimal

effect on the calculated cloud dimensions.

Errors Due To Limited Visibility Of The Exhaust Cloud

For Missions 41C and 51A, the cloud is obscured in one or more views either by bright, light-saturated portions of the film (UCS 6, 41D), environmental clouds (UCS 2, 51A), the cameras' limited fields of view (UCS 9, 41D; UCS 6, 51A), or an unusable film (UCS 9, 51A). For 41D, the severity of these problems was lessened by the availability of other views. Estimations were made for the measurements of the cloud when it was partially obscured (which only occurred for a few of the tracings). Since these estimations agreed well with information from the other views, they were not disregarded. For Mission 51A, however, only the UCS 6 view was usable for cloud measurements. After T + 7:00, though, the cloud began to extend beyond the camera view, rendering further measurements inaccurate.

The limited cloud visibility for Mission 51A also decreased the accuracy of all range calculations which, in turn, affected the calculated cloud dimensions. The straight-line range approximation employed for 51A was only accurate to within about 200 m because the entire cloud was not visible in both views. Moreover, since the cloud in the UCS 6 view spread out horizontally (in the direction of travel), the points selected for measurements may have been up to 500 m away from the center point corresponding to the calculated range. This can cause an error of about 250 m. Since there were no identifiable ground features to use as position references, it was impossible to accurately compensate for these range differences. Therefore these range uncertainties probably caused errors in the cloud dimensions of up to 300 m for Mission 51A.

Errors Due To The Exhaust Cloud's Non-Uniform Shape

Since the exhaust cloud was non-uniform in shape and quite wide near the top for all launches, it was difficult to determine if the cloud top seen on the films was the actual maximum height or a lower height closer to the camera. Since the cameras were located relatively close to the cloud and on the ground, the vertical angle (σ) to the cloud top was 20-30 degrees. If the slope of the top part of the cloud from its sides to its center maximum is less than the vertical angle, the absolute maximum will not be seen on the film. In addition, because the cloud appeared to be rather non-uniform in shape and the camera sites are not the same distance away, the altitude of the apparent cloud top may differ from view to view. Thus, the shape of the cloud not only causes errors in the calculated heights, but also creates discrepancies between views.

Since an error of 400 m in the range will cause an error of up to 200 m in the cloud height, it is important to determine the correct range to the point selected as the cloud top. However, the cloud shape makes accurate range determinations difficult. For example, (see Figure 10) if the range to the top is taken to be where the cloud center is located and the absolute top is not visible, the calculated height will be too large. In contrast, if the apparent cloud top is estimated and the range calculated for that point, the altitude will be too low (since the apparent top is lower than the actual top).

Finally, because of the size of the exhaust cloud, it is virtually impossible to locate a particular cloud point in two different camera views. Therefore, determining the ranges is only accurate to about 500 m. The combination of errors due to the cloud shape probably causes the calculations of the cloud tops and widths to have errors of up to 250 m for all launches.

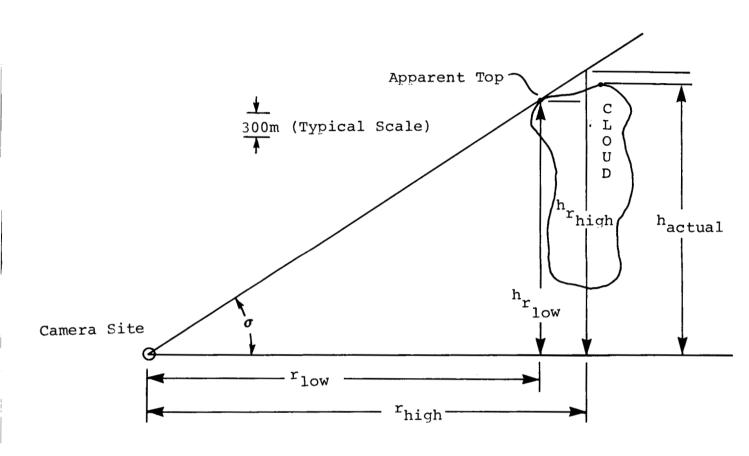


Figure 10. Errors caused by cloud asymmetries and large widths

SECTION 7 - REFERENCES

Holle, R. L., 1982. Photogrammetry of thunderstorms. <u>Thunderstorms: A Social, Scientific and Technological Documentary</u>, E. Kessler, Ed. U.S. Department of Commerce.

APPENDIX A

SHUTTLE EXHAUST CLOUD MEASUREMENTS

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16. ABSTRACT			1
	togrammetry are used		
			edy Space Center, Florida 6 mm camera recorded the
			A NASA timing code (IRIG)
			each frame. Measurements
•	-		nterest which were traced
onto paper at one minu			
measurements were conve			
			x width at top, average
width, and base are pre			
used to check results of			
position of the rocket			iew, and the known
dimensions of buildings	s appearing in selec	ed film frames.	
For each of the th	ree launch days, a i	emperature invers	ion was present and the
cloud responded by firs	st rising into the in	version then dese	nding to about the
inversion height. Max			
2200 m for Mission 41C			
between 700 and 1000 ${\tt m}$	after 10 minutes.	verage widths ran	ged from 500 to 1700 m
depending on mission ar	nd camera. Photograp	hs of digitizatio	ns of selected film
		he irregular clou	d shapes. Error sources
for this analysis are a	also discussed.		
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